

#### WING FILLET

### APPLICABLE MODELS

PA-24, PA-24-250, PA-24-260 and PA-24-400

### WING FILLET

### INSTALLATION MANUAL

### ISSUE DATE 04/01/95

Knots 2U, Ltd. 709 Airport Road Burlington, WI 53105 (262) 763-5100 E A A APPROVED

OCT 3 1 1995 John Act - 116C CHICAGO AIRCRAFT CERTIFICATION OFFICE CENTRAL REGION

REVISIONS							
REV# DATE	PAGE	EFFECT					
A 10/25/95	1&2	ADDED RUBBER CHANNEL AND IMPLEMENTED RETENTION OF TOP FLAP/FUSELAGE SEALS IF ALREADY INSTALLED AND INSTALLATION OF SAME, IF NOT ALREADY INSTALLED.					

PA-24

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### WING FILLET

## STC No. SA371CH

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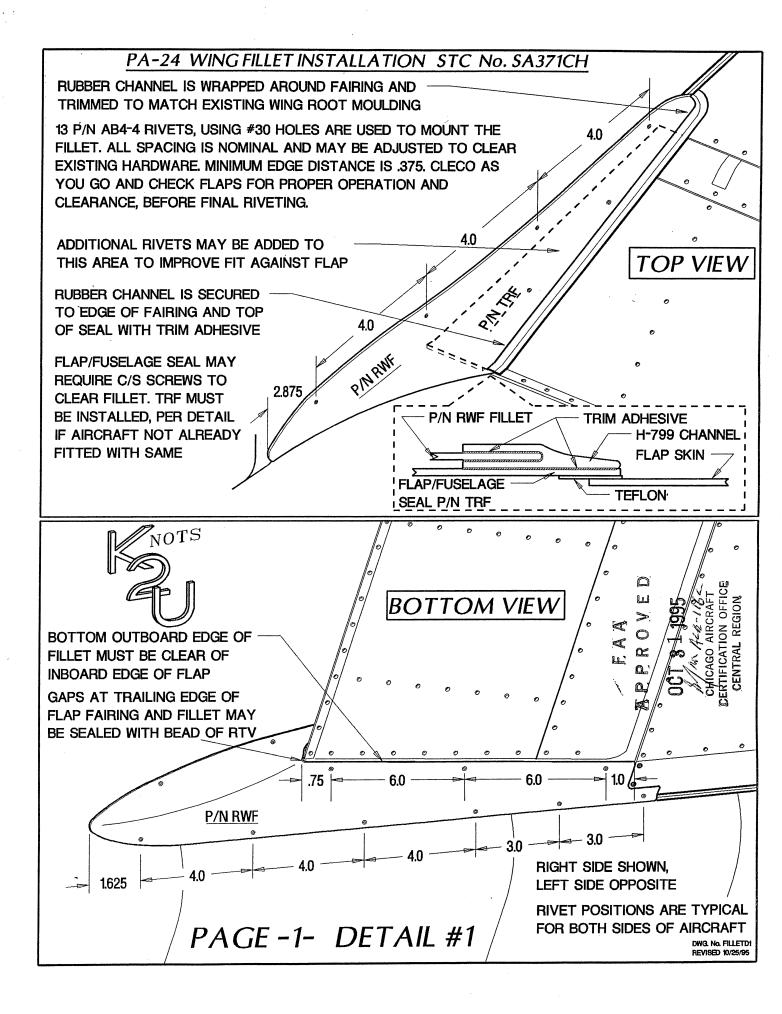
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#### WING FILLET

STC No. SA371CH

#### = SECTION 2.0 PARTS LIST =

FAA APPROVED

PART NUMBER	<u>NO. REQ</u>	DESCRIPTION					
RWF	1	RIGHT WING FILLET	OCT 31 1995				
LWF	1	LEFT WING FILLET	CHICAGO AIRCRAFT				
AB4-4	30	RIVET	CERTIFICATION OFFICE				
H-799	2.5 FT.	CHANNEL (WAG-AERO CA	TALOGUE)				
TRF*	1*	TOP RIGHT FLAP/FUSELAG	GE SEAL*				
TLF*	1*	TOP LEFT FLAP/FUSELAGE	SEAL*				
* (NOT SUPPLIED IE ALREADY INSTALLED ON AIRCRAET)							

(NOT SUPPLIED IF ALREADY INSTALLED ON AIRCRAFT)

### <u>= SECTION 3.0 PAPERWORK =</u>

PERFORM PAPERWORK (337 AND LOG BOOK ENTRIES). PLACE SUPPLEMENTAL TYPE CERTIFICATE AND KNOTS 2U, INC. MAINTENANCE MANUAL WITH LOG BOOKS.

WEIGHT AND BALANCE CHANGE IF P/N TRF ALREADY INSTALLED = NEGLIGIBLE WEIGHT AND BALANCE CHANGE IF P/N TRF ADDED = +1.0 LB. ARM = 140 IN. MOMENT = 140.0

= <u>SECTION 4.0 MAINTENANCE MANUAL</u> =

K<u>NQTS2U,INC</u>.

PIPER PA-24

#### WING FILLET

#### PART A. INSPECTION

1. DURING ANNUAL OR 100 HOUR INSPECTIONS, INSPECT FIBERGLASS FOR CRACKS AND ATTACHMENT HARDWARE FOR LOOSENESS OR WEAR. ALSO, CHECK FLAP OPERATION FOR BINDING.

#### PART B. MAINTENANCE

- 1. IF THE ATTACHMENT HARDWARE IS FOUND TO BE EXCESSIVELY WORN OR LOOSE DURING THE 100 HOUR/ANNUAL INSPECTION, IT SHOULD BE REPLACED.
- 2. IF THE FLAP BINDS OR THE FAIRING IS NOT SEALING PROPERLY THE FAIRING MUST BE ADJUSTED AND/OR TRIMMED TO CORRECT THE DEFECT.

#### PART C. CRACKING OR DEFECTS

1. IF A CRACK IS FOUND IN THE FILLET, REPAIR THE CRACK ACCORDING TO FAR 43.13 1A ACCEPTABLE METHODS, TECHNIQUES, AND PRACTICES AIRCRAFT INSPECTION AND REPAIR, CHAPTER 2.