# **APPLICABLE MODELS**

PA-30 and PA-39

WING FILLET

INSTALLATION MANUAL

ISSUE DATE 04/01/95

Knots 2U, Ltd. 709 Airport Road Burlington, WI 53105 (262) 763-5100

<u>REVISIONS</u>						
REV# DATE PAGE	EFFECT					
A 10/25/95 1&2	ADDED RUBBER CHANNEL AND IMPLEMENTED RETENTION OF TOP FLAP/FUSELAGE SEALS IF ALREADY INSTALLED AND INSTALLATION OF SAME, IF NOT ALREADY INSTALLED.					
	APPROVED					
·	OCT 3 1 1995  M/M 196-116C  CHICAGO AIRCRAFT					

CENTRAL REGION

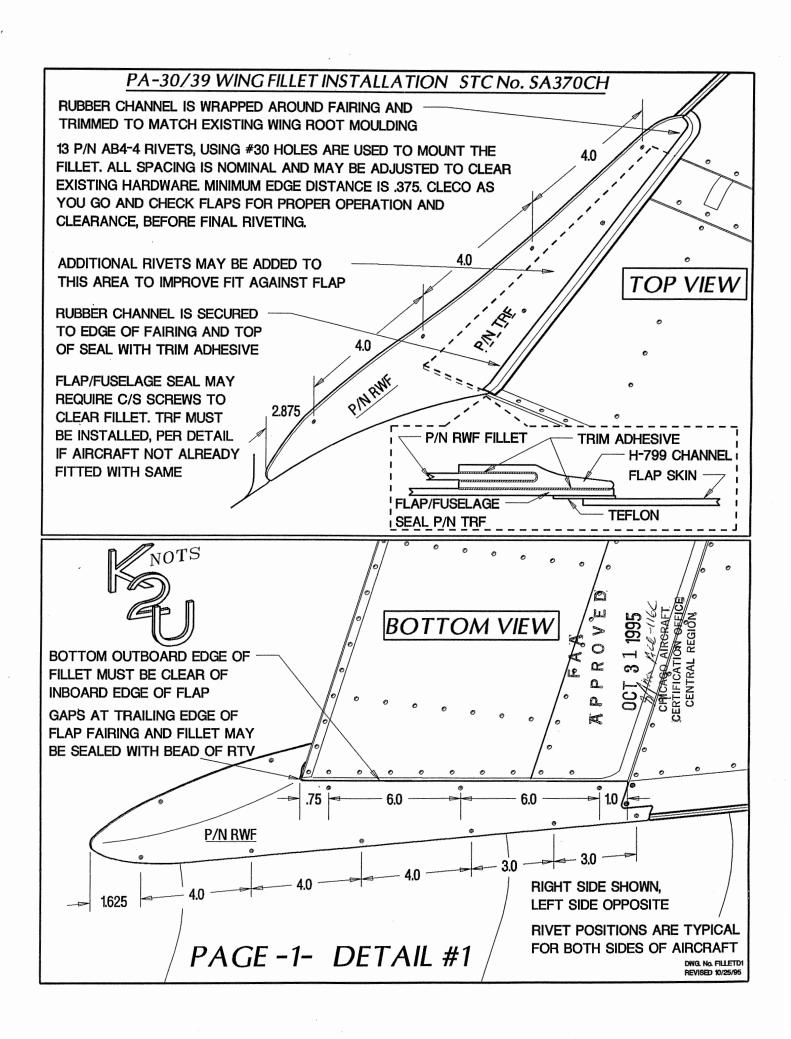
# TABLE OF CONTENTS

SECTIO	N No.	PAGE No.
1.0	-DETAIL 1	1
2.0	-PARTS LIST	2
3.0	-PAPERWORK	2
4.0	-MAINTENANCE MANUAL	2
5.0	-SUPPLEMENTAL TYPE CERTIFICATE	3

FAA APPROVED

OCT 3 1 1995

CHICAGO AIRCRAFT
CERTIFICATION OFFICE CENTRAL REGION



PA-30/39	WING FILLET	STC No. SA370CH

F. A.A

APPROVED

## = SECTION 2.0 PARTS LIST =

PART NUMBER	NO. REQ	DESCRIPTION	OCT 3 1 1995 Am Aca 1160	
RWF	1	RIGHT WING FILLET	CHICAGO AIRCRAFT	
LWF	1	LEFT WING FILLET	CERTIFICATION OFFICE CENTRAL REGION	
AB4-4	30	RIVET	OLIVINAL REGIONS	
H-799	2.5 FT.	CHANNEL (WAG-AERO CATALOGUE)		
TRF*	1*	TOP RIGHT FLAP/FUSELAC	GE SEAL*	

TLF\* 1\* TOP LEFT FLAP/FUSELAGE SEAL\*

\*(NOT SUPPLIED IF ALREADY INSTALLED ON AIRCRAFT)

## = SECTION 3.0 PAPERWORK =

PERFORM PAPERWORK (337 AND LOG BOOK ENTRIES). PLACE SUPPLEMENTAL TYPE CERTIFICATE AND KNOTS 2U, INC. MAINTENANCE MANUAL WITH LOG BOOKS.

WEIGHT AND BALANCE CHANGE IF P/N TRF ALREADY INSTALLED = NEGLIGIBLE

WEIGHT AND BALANCE CHANGE IF P/N TRF ADDED =  $\pm 1.0$  LB. ARM = 140 IN. MOMENT = 140.0

## = SECTION 4.0 MAINTENANCE MANUAL =

#### KNOTS 2U, INC.

PIPER PA-30 & PA-39

### WING FILLET

#### PART A. INSPECTION

1. During annual or 100 hour inspections, inspect fiberglass for cracks and attachment hardware for looseness or wear. Also, check flap operation for binding.

### PART B. MAINTENANCE

- 1. IF THE ATTACHMENT HARDWARE IS FOUND TO BE EXCESSIVELY WORN OR LOOSE DURING THE 100 HOUR/ANNUAL INSPECTION, IT SHOULD BE REPLACED.
- 2. IF THE FLAP BINDS OR THE FAIRING IS NOT SEALING PROPERLY THE FAIRING MUST BE ADJUSTED AND/OR TRIMMED TO CORRECT THE DEFECT.

#### PART C. CRACKING OR DEFECTS

1. IF A CRACK IS FOUND IN THE FILLET, REPAIR THE CRACK ACCORDING TO FAR 43.13 1A ACCEPTABLE METHODS, TECHNIQUES, AND PRACTICES AIRCRAFT INSPECTION AND REPAIR, CHAPTER 2.