

SEMI-TAPERED WING MODEL GAP SEALS	Issue Date: 09/06/90
STC No. SA1521GL	Rev. F
Manual No. 32FGS-M	Rev Date: 02/05/10

FLAP, AILERON FLAP/FUSELAGE AND STABILATOR GAP SEALS INSTALLATION AND MAINTENANCE MANUAL

Aircraft Eligibility:

PA-32R-301, PA-32R-301T, PA-32-301, PA-32-301T, PA-32R-301(SP), PA-32R-301(HP),
PA-32-301FT, PA-32-301XTC

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Revision Control

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Section 1.0

Introduction

This manual describes the installation of gap seals to the aircraft's flaps, ailerons, and stabilator trim tab.

PERFORMANCE FLIGHT TESTING.

1. The rigging of the aircraft should be checked before the installation, and any changes to the rigging should be performed before the flight testing.
2. The flight testing must be done with the aircraft loaded to the same gross weights, and as far as possible under the same atmospheric conditions. Early morning or late evening will usually provide the smoothest air.
3. Under identical conditions, a good indication of performance change is the "indicated" airspeed. Example, if you climb to 5,000 feet and find after stabilizing at maximum cruise you have 24 inches of manifold pressure, the altitude that gives you 24 inches of manifold pressure on the after flights will be close to the same density altitude as the early flights. If the air pressure is creating an identical manifold pressure reading, and the air speed indicator is reading higher, it is obvious airspeed has increased. This is because the airspeed indicator is a pressure reading device. A density altitude indicator will help to verify the same flight conditions.
4. Another method is timing across 2 parallel roads between which you know the exact distance. (The further apart, the less chance for error) Stabilize the aircraft perpendicular to the roads and fly straight across the first. Punch the timer at the exact point where the leading edge of the wing crosses the road and maintain a constant heading at 90 degrees to the road. After maintaining constant altitude and power settings, punch the clock again when the leading edge crosses the second road. Repeat the procedure back and forth several times and record all resulting times. Calculate the ground speed for each flight in all directions and average the number of flights. Repeat after the mod installation.

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Section 2.0

Left and Right Aileron Gap Seal Installation

NOTE #1:

This manual describes the installation of the Knots 2U Flap, Aileron, Flap/Fuse & Stabilator seal kit. The appropriate Piper service manual for your aircraft should be used in conjunction with this manual. The best time to paint the parts is before installing, or after all holes are drilled and before final installation. Priming is not necessary on the aluminum parts as they are already primed with a zinc chromate wash primer. Surfaces should be "scuffed" with a scotch brite and painted. Cover the chaff strip on the trailing edge of the seals with masking tape before painting.

2.1 Disconnecting Left Aileron and Removing Rivets

Remove actuator arm bolt from left aileron. The aileron will now swing down enough to allow installation of the gap seals. Remove the (45) rivets from the upper wing skin at the trailing edge in front of the aileron per (DETAIL #1), cleco as you go.

2.2 Installing P/N 32FIA Inboard Aileron Gap Seal

Insert the inboard aileron gap seal P/N 32FIA between the top wing skin and the stiffener (reference DETAIL #1) Slide the seal into the wing 1.0" with the inboard edge of the seal flush with the inboard edge of the aileron. The outboard end of the seal should be flush with the actuator arm cutout; seal may be trimmed to fit. Make sure the teflon side of the seal is facing the aileron. With the seal held firmly in place, and using the holes just drilled as a guide, drill #30 holes in seal; cleco as you go.

2.3 Installing P/N 32FOA Outboard Aileron Gap Seal

Insert the outboard aileron gap seal P/N 32FOA between the top wing skin and the stiffener (reference DETAIL #1) Slide the seal into the wing 1.0" with the outboard edge of the seal flush with the outboard edge of the aileron. The inboard end of the seal should be flush with the actuator arm cutout; seal may be trimmed to fit. Make sure the teflon side of the seal is facing the aileron. With the seal held firmly in place and using the holes just drilled as a guide, drill #30 holes in seal, cleco as you go.

2.4 Checking Aileron Travel and Final Cleanup

Rotate aileron through its full travel. Check the amount of friction between the aileron seal and the aileron. The seal should rest lightly on the aileron in the neutral position. If the seal is placing too much pressure on the aileron, or not resting on the aileron, use a straight block of wood to push the seal either up or down as required. When the aileron is rotated up it is natural for the aileron to pull away from the seal. Mark any locations where rivets on the aileron will interfere with the aileron seal. Some rivets around the actuator arm cutout, or at the ends of the aileron may be removed and flush rivets installed. For other rivets, using a pliers carefully bend the seal up enough to prevent the seal from catching on aileron. When satisfactory, remove all aileron seals. Clean shavings from aircraft and seals. De-burr and corrosion proof all holes with alodine or equivalent. Seals may be painted at this time.

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2.5 Riveting Aileron Seals

Rivet all seals in place using AN470AD-4-4 rivets.

2.6 Re-Connecting Aileron

Reconnect aileron actuating rod. Check for smooth and full travel of the aileron from the cockpit.

2.7 Right Side Installation.

Repeat steps 2.1 thru 2.6 on right side of aircraft.

2.8 Removal of Aileron Seals.

If a seal needs to be removed it can be accomplished by disconnecting the aileron per section 2.1 and drilling the attachment rivets out. Care should be taken to not enlarge the attachment holes while drilling.

Section 3.0

Left and Right Flap Gap Seal Installation

NOTE #2:

If Knots 2U flap hinge fairings are being installed on the aircraft, some labor savings can be accomplished by installing them before the flap gap seals. If rigging of the flaps is necessary it should be completed before installation of flap seals.

NOTE #3:

Some later model PA-32R-301(SP) and PA-32R-301(HP) models have a flap seal on the plane installed by Piper. Piper lengthened the lower wing skin to come back almost to the flap closing up the gap. If your aircraft has this our flap gap seal kit will not fit and should be left off.

3.1 P/N LIF, Left Inboard Seal Locating And Drilling.

(DETAIL #1) Per the appropriate Piper Service Manual, fully lower flaps. At the inboard end remove the clevis bolt and spacer from the flap actuating rod terminal end. Retain parts for re-installation. The flap will now swing down and allow room for installation. Position P/N LIF with the end tab against the fuselage and 5/8" flange against rear of the wing. Position the bottom of the seal flush with the bottom of the wing skin. There are (2) existing rivets on the fuselage where the P/N LIF meets it. These rivets may be drilled out and the holes used to attach P/N LIF. (If these holes do not allow enough edge distance, drill (2) new holes while observing edge distances) With the seal properly located using the pilot holes, mark and drill the (2) outboard holes in the rear wing spar to a #40 hole size. Cleco seal in place.

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3.2 P/N OF & LOF, Outboard Flap Seal Locating and Drilling.

(DETAIL #1) Using 1/32" edge spacing, place (1) P/N OF 1/32" outboard of P/N LIF just installed. Position the bottom of the seal flush with the bottom of the wing skin. Drill clearance holes in the seal flange to allow the seal to clear any existing rivets that prevent the seal from laying flush against the spar. Drill # 40 holes through pilot holes and cleco seal in place. Repeat procedure for next outboard seal P/N LOF.

3.3 Enlarging Holes.

Enlarge attachment holes in wing to a #27 hole size. Also enlarge the pilot holes in the P/N's OF, LOF, and LIF to a #27 hole size. De-burr all holes in gap seals and airframe, remove all shavings from aircraft and corrosion proof all holes with Alodine or equivalent. The seals may be painted at this time.

3.4 Final Installation of Flap Seals.

(DETAIL #1) With gap seals temporally held in place with PK screws, move flap up and down and check for proper contact and alignment with seals. With flap in full up position the seals should make light contact with flap. If seals are not making contact they can be moved up slightly by re-bending them. If the seals are too high, they may be bent down slightly using a straight edge. Rivet flap seals in place using (56) P/N CR3243-4-2 Cherry Max Rivets.

3.5 Re-Connecting Flap.

Per the appropriate Piper Service Manual, reinstall clevis bolt and spacer in flap actuating rod terminal end. Run flap up and down several times to check seal for any binding and proper alignment.

3.6 Right Side Installation.

Repeat steps 3.1 thru 3.5 on right wing. P/N LOF will be replaced by P/N ROF and P/N LIF will be replaced by P/N RIF. All other parts are the same.

3.7 Removal of Flap Seals.

If a seal needs to be removed it can be accomplished by lowering the flap per section 3.1 and drilling the attachment rivets out. Care should be taken to not enlarge the attachment holes while drilling.

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Section 4.0

Aileron and Flap Seal Paperwork

Perform paperwork (337, log book entry and Weight & Balance change). Place this manual and Supplemental Type Certificate with aircraft log books.

Kit	Weight in lbs	Arm in inches
Aileron Gap Seals	0.32	129
Flap Gap Seals	1.00	129

Section 5.0

Aileron and Flap Seal Parts List

Part Number	Qty	Description
32FIA	2	Freise Aileron Seal Inboard
32FOA	2	Freise Aileron Seal Outboard
OF	2	Flap Seal
LOF	1	Left Outboard Flap Seal
ROF	1	Right Outboard Flap Seal
LIF	1	Left Inboard Flap Seal
RIF	1	Right Inboard Flap Seal
AN470AD-4-4	90	Roundhead Rivet
CR3243-4-2	56	Roundhead Cherry-Max Rivet

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Section 6.0

Left and Right Flap / Fuselage Gap Seals

NOTE #4: If rigging of the aircraft is necessary, it should be accomplished before the installation of the flap / fuselage gap seals.

6.1 P/N LFF, Left Flap / Fuselage Gap Seal Installation.

(DETAIL #1) With Flap full up, place P/N LFF, (Left Flap / Fuselage Gap Seal) on top of inboard end of flap against fuselage so the seal covers the space between the inboard edge of the flap and the fuselage. Position so that the forward end of the curved portion is even with the trailing edge of the flap, and so it makes solid contact with the top of the flap surface. If rivets interfere with the seal it may be trimmed to clear. With seal properly located, and observing all edge distances, mark (3) hole locations, (1) at each end of the seal flange and (1) in the center. If existing hardware falls under a hole location, the hole may be relocated. Drill #40 hole locations at these points and temporarily attach seal using (3) P/N TRA 4X1/4 screws. Lower and raise the flap several times to confirm proper contact and no binding. Mark the seal where it overhangs the bottom of the aircraft. Excess material should be trimmed off and the seal de-burred. Enlarge holes to a #27 hole size. De-burr and corrosion proof holes and trimmed edges with Alodine or equivalent. Seal may be painted at this time. Rivet seal in place using (3) P/N CR3243-4-2 Cherry Max rivets or equivalent.

6.2 P/N RFF, Right Flap / Fuselage Gap Seal Installation.

Repeat step 6.1 on right side of aircraft.

6.3 Removal of Flap / Fuselage Seals.

If a seal needs to be removed it can be accomplished by lowering the flap and drilling the attachment rivets out. Care should be taken to not enlarge the attachment holes while drilling.

6.4 Paperwork

Perform paperwork (337, log book entry and Weight & Balance change). Place this manual and Supplemental Type Certificate with aircraft log books.

Kit	Weight in lbs	Arm in inches
Flap / Fuselage Seals	0.14	129

6.5 Flap / Fuselage Seal Parts List

Part Number	Qty	Description
LFF	1	Left Flap / Fuselage Seal
RFF	1	Right Flap / Fuselage Seal
CR3243-4-2	6	Cherry Max Rivet

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Section 7.0

Stabilator Gap Seal Installation

Note #5. It is required that the stabilator be balanced after installation and painting of the seal is complete. Reference the proper service manual for balancing procedures. If stabilator will be balanced off the plane it is easier to accomplish the installation after stabilator has been removed.

7.1 Disconnecting Trim Tab Actuator Arm.

Remove the (4) bolts connecting the Stabilator trim tab actuator arm. Move the left and right tabs upward as far as free travel allows.

7.2 Rivet Removal.

Remove existing #4 rivets from the lower rear spar of the Stabilator.

7.3 Installing Stabilator Gap Seal

(DETAIL #2) Insert gap seal P/N T28 on left side between lower spar flange and lower Stabilator skin. Position the seal 5/8 inch (nominal) inward between spar and skin with chafe strip aft and facing up toward trim tab. Seal should be positioned so that there is a 1/8 inch gap between the inboard edge of the seal and the trim tab actuator base. Seal may then be trimmed on the outboard end so that it does not extend further outboard than the trim tab. Lower trim tab so it rests against seal. At the very outboard end of seal some models have a roundhead rivet on the trim tab that will interfere with flush fit of Stabilator Seal. This rivet should be removed and replaced with a countersunk rivet of the same size. Repeat procedure on right side.

7.4 Drilling Rivet Holes.

With seal in proper location and using rivet holes in skin as a guide, drill # 30 holes in P/N T28 Stabilator Seal, Cleco as you go. With seal drilled and clecoed, lower trim tab and confirm proper contact of seal against trim tab. With the trim tab in the neutral position it should lightly contact the seal. Move trim tab through its full travel and confirm there is no excess binding. If binding occurs or seal does not meet with the trim tab in the neutral position, the seal may be bent slightly up or down using a straight edge.

7.5 Final Cleanup.

Remove the gap seal from aircraft and clean all drill shavings from area. De-burr all holes and trimmed areas, corrosion proof holes and trimmed edges with Alodine or equivalent. Seal may be painted at this time.

7.6 Riveting P/N T28 Stabilator Seal.

Cleco gap seals in place. Rivet every other hole using AN 470AD-4-4 rivets, while leaving every other cleco in place. After 1/2 of the rivets are in place, remove the other clecos and finish riveting seals.

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7.7 Re-Balancing Stabilator.

It is the responsibility of the installer to assure that the stabilator is properly balanced per the appropriate Piper Service Manual. If any painting is done to the seals or stabilator, it must be done before balancing. Check for full travel after balancing and that any weights added to balance arm do not restrict stabilator travel. Trim tab actuator arm should be reconnected and the stabilator run through its full travel to check for binding.

7.8 Removal of Stabilator Seal.

If a seal needs to be removed it can be accomplished by drilling the attachment rivets out. Care should be taken to not enlarge the attachment holes while drilling. Reference Section 7 of this manual for details.

7.9 Paperwork.

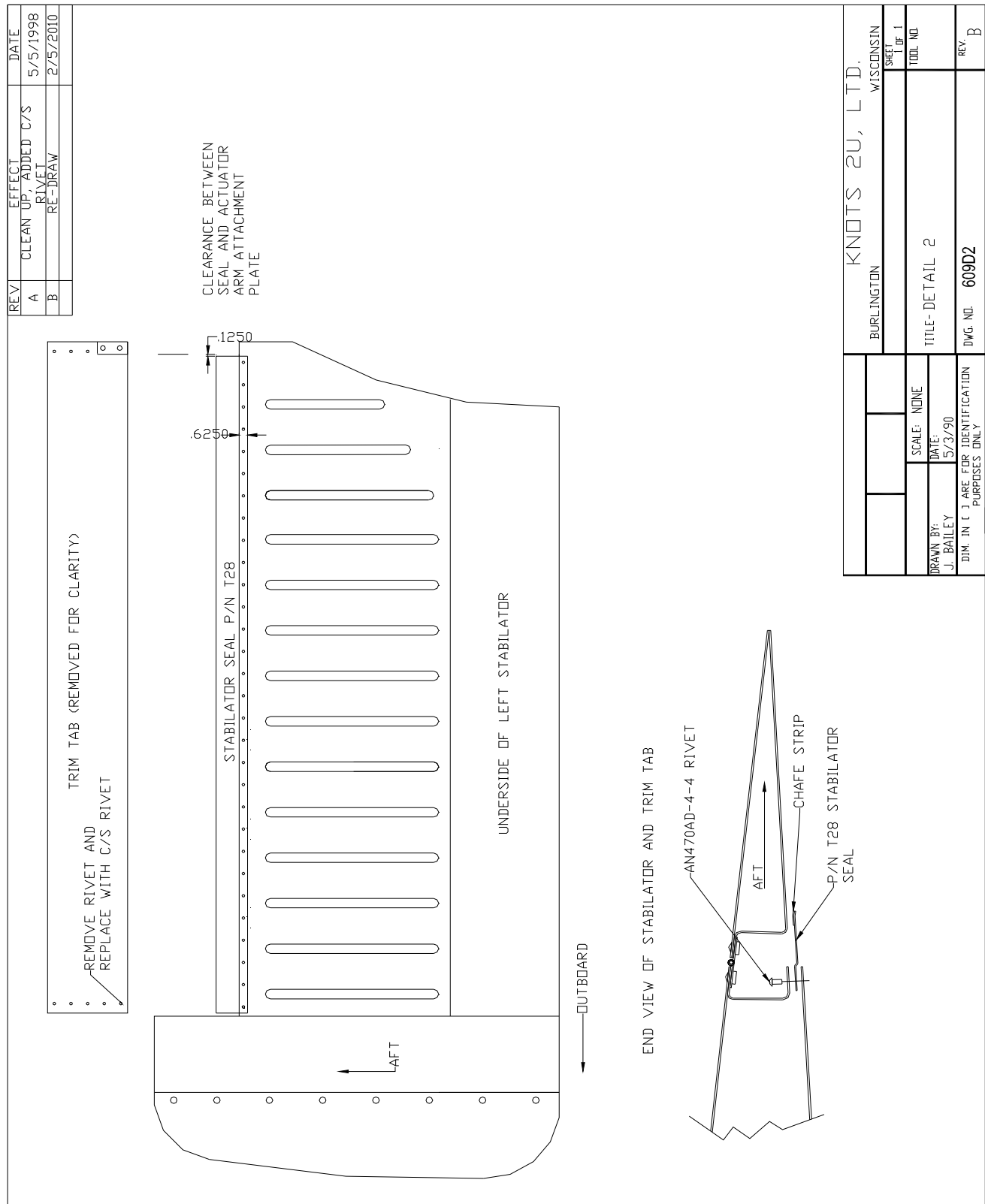
Perform paperwork (337, log book entry and Weight & Balance change). Place this manual and Supplemental Type Certificate with aircraft log books.

Kit	Weight in lbs	Arm in inches
Stabilator Gap Seals	0.42	Per appropriate Piper Service Manual

7.10 Stabilator Seal Parts List

Part Number	Qty	Description
T28	2	Stabilator Gap Seal
AN470AD-4-4	67	Rivet

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Section 10.0

Maintenance / Instructions for Continued Airworthiness

The Airworthiness Limitations section is FAA approved and specifies maintenance required under Secs. 43.16 and 91.403 of the Federal Aviation Regulations unless an alternative program has been FAA approved.

For the current copy of this manual please contact Knots 2U, Ltd. at 262 763-5100 or via email at technical@knots2u.com.

PART A. INSPECTION

1. Daily inspection at preflight to ensure there is no binding of controls, bent gap seals, abrading of rivets or control surfaces, or broken parts.
2. When aircraft has been stored outside during snow or freezing conditions, a careful inspection should be made of the areas behind and under the seals for ice accumulations. If ice is found, which cannot be removed by careful brushing, the aircraft should be de-iced.
3. 100 hour inspections are suggested to check for abrading of rivets, control surfaces, and seals. Chafe tape should be inspected for peeling or excessive wear. Check all hardware and attachment of all seals.

PART B. MAINTENANCE

1. There are no special tools required to maintain the seals. Any tools needed are basic hand tools.
2. Keep the seal surface clean of oil and dirt, and the edge of the seal touching the control surface smoothly. If the Gap Seal appears to be abrading the control surface, 3M 5490 Teflon Tape, or equivalent, may be applied to the Gap Seal to act as a wear surface. The tape should be applied before further flight to prevent control wear.
3. If upon installation, or through wear, there is a warp in the seal or it lies unevenly, you may drill a #40 hole and cut the seal in a direction 90 degrees from the trailing edge. Drill the hole in the center of the warp, 1/2 inch from the trailing edge of the surface that the seal is attached to. The cut in the seal should be trimmed to give a slot 1/16th" wide, with parallel edges. The slots should be no closer than 6 inches from each other or the end of the seal. Refer to detail MM Detail, next page.
4. When aircraft is painted, care should be taken to prevent paint, paint remover, or solvents from contacting the Chafe Strip. If Chafe Strip is damaged, refer to PART B Paragraph 2 of this manual for replacement specifications.

PART C. BALANCING

1. If any change is made to the horizontal stabilator; if chafe strip is added, if slots are cut to relieve warping, or if the seals are painted, the stabilator must be rebalanced.

PART D. CRACKING, DEFECTS, LOOSE RIVETS.

1. If cracks are found in a gap seal, stop drill the crack. If there are more than 3 cracks in a gap seal, the seal must be replaced.
2. If the chafe strip peels, 3M 5490 Teflon Tape, or equivalent, may be applied
3. If there are excessive bends or kinks in the seal, and the airflow over the control surface is disturbed, the seal must be replaced.
4. If seal rivets become loose you may drill the rivets and replace with the next size rivet.

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